

# Scatter Working Group

## Test Case 1



Version 3  
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- **Geometry is available here: (it is very strongly desired to use the provided IGES file in the ONERA OAT15A zip file and not the raw coordinates)**

<https://aiaa-dpw.larc.nasa.gov/geometry.html>

- **Committee-supplied RANS grids are available here**

<https://aiaa-dpw.larc.nasa.gov/grids.html>

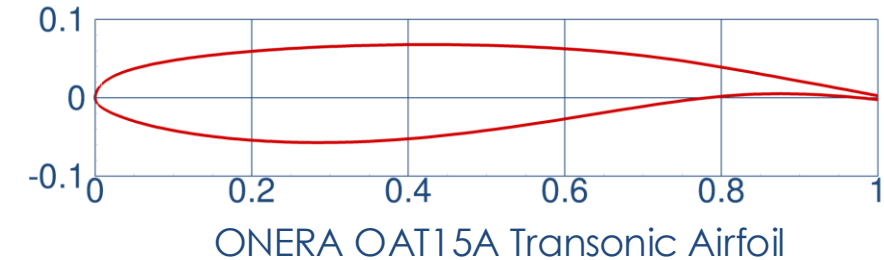
- **Experimental data are available here**

<https://aiaa-dpw.larc.nasa.gov/experiment.html>

# Test Case 1: Workshop-Wide Validation

- **Validation of steady CFD analysis, required**
- **Users are encouraged to employ best practices**
- **Settings**
  - Steady CFD (e.g., RANS)
  - Prefer some version of SA, multiple turbulence models can be submitted
  - Use periodic boundary conditions for sidewall boundary conditions
- **Grids**
  - Six-member grid family; four are required, six are desirable
  - Encourage use of committee-supplied grids; user-generated grids are acceptable
  - Three committee-supplied once-cell-wide grid topologies are provided
- **Conditions**
  - Mach 0.73,  $Re_c=3m$  (based on chord length),  $T_{static}=271$  K (487.8 R)
  - Alpha: 1.36, 1.50, 2.50, 3.00, 3.10
  - Experimental conditions (for reference):  $P_{total}=102.4$  kPa;  $P_{static}=71.8$  kPa

Jaquin, et al. "Experimental Study of Shock Oscillation over a Transonic Supercritical Profiles." AIAA Journal, Vol. 47, No. 9, 2009. Pages 1985-1994.



# Test Case 1: Data Submission

- **Please follow these instructions**

<https://aiaa-dpw.larc.nasa.gov/postprocessing.html>

- **Required data**

- Forces and Moments

DPW8-AePW4\_ForceMoment\_v4.dat

- Surface cuts

DPW8-AePW4\_SectionalCuts\_v4.dat

Use sectionalCutter-v1.mcr

- Convergence data

DPW8-AePW4\_Convergence\_v4.dat

- Contour plots

Use airfoilImages-v1.mcr

- Boundary layer profile data (in work)



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