

# Structured Overset Grids for the NASA Common Research Model (CRM)

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## **Gridding Guidelines**

Initial spacing normal to all viscous walls (RE = 5M based on  $C_{REF}$ =275.80"):

- Coarse:  $y^+ \sim 1.0$   $\Delta y = 0.001478$  in
- Medium:  $y^+ \sim 2/3$   $\Delta y = 0.000985$  in
- Fine:  $y^+ \sim 4/9$   $\Delta y = 0.000657$  in
- Extra-Fine:  $y^+ \sim 8/27$   $\Delta y = 0.000438$  in

Total grid size to grow ~3X between each grid level for grid convergence cases

• For structured meshes, this growth is ~1.5X in each coordinate direction

Growth rate of cell sizes in the viscous layer should be < 1.25

Farfield located at ~100 C<sub>REF</sub>'s for all grid levels

For the Medium Baseline Grids:

- Chordwise spacing for wing and tail leading edge (LE) and trailing edge (TE) ~0.1% local chord
- Wing and tail Spanwise spacing at root ~0.1% local semispan
- Wing and tail Spanwise spacing at tip ~0.1% local semispan
- Cell size near fuselage nose and after-body ~2.0% CREF

Wing and Tail Trailing Edge Base:

- Minimum of 8 cells across TE base for the coarse mesh
- Minimum of 12 cells across TE base for the medium mesh
- Minimum of 16 cells across TE base for the fine mesh
- Minimum of 24 cells across TE base for the extra-fine mesh

Be multi-grid friendly





Suggested Grid Size for the NASA CRM Wing-Body:

• Medium = ~8M cells/nodes

Suggested Grid Sizes for the NASA CRM Wing-Body-Tail ( $i_H = 0^\circ$ ):

- Coarse = ~3.5M cells/nodes
- Medium = ~10M cells/nodes
- Fine = ~35M cells/nodes
- Extra-Fine = ~100M cells/nodes





First built the medium grid following established "best practices" for overset grid generation

- Chan, Gomez, Rogers, Buning, "Best Practices in Overset Grid Generation", AIAA 2002-3191
- Vassberg, DeHaan, Sclafani, "Grid Generation Requirements for Accurate Drag Predictions Based on OVERFLOW Calculations", AIAA 2003-4124

Medium Grid Generation Process



Built coarse, fine and extra-fine grids using the medium volume grids

- Used an in-house program called P3D\_REDIM
- Re-ran PEGASUS5 and MIXSUR





#### Structured Overset Grid Systems

➤ 11 zones for Wing-Body

> 17 zones for Wing-Body-Horizontal

Medium grid is typical for drag quality design studies

Wing-Body

		1/N <sup>2/3</sup> x 10 <sup>5</sup>		-	Constant	Growth
Grid	Points		1 <sup>st</sup> Cell Size	Ут	Cells	Rate
Medium	12,267,995	1.88	.00079 in	.66	3	1.19

#### Wing-Body-Horizontal

Grid	Points	1/N <sup>2/3</sup> x 10 <sup>5</sup>	1 <sup>st</sup> Cell Size	y+	Constant Cells	Growth Rate
Coarse	7,221,233	2.68	.00104 in	.87	2	1.26
Medium	16,932,913	1.52	.00079 in	.66	3	1.19
Fine	56,531,489	0.68	.00052 in	.44	4	1.12
Extra Fine	189,413,153	0.30	.00035 in	.29	6	1.08





# Wing-Body-Horizontal Surface Abutting Grids

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#### Volume Grids – Coarse Grid Shown Here



#### Box Grids at the Symmetry Plane





AIAA DPW-IV

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#### **Body Grids**



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#### Wing-Body Collar Grids





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#### Wing Grids





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coarse

cells .75

medium

24 cells

fine

36 cells

1.5 x m

extra-fine

54 cells 1.5 x f

<u>}18</u>

x m

## Wing Trailing-Edge Cap Grids

grid	J	к	L	total
coarse	<b>73</b>	<b>37</b>	<b>37</b>	99,937
	.75 x m	.75 x m	.75 x m	.43 x m
medium	97	49	49	232,897
fine	<b>145</b>	<b>73</b>	<b>73</b>	<b>772,705</b>
	1.49 x m	1.49 x m	1.49 x m	3.32 x m
extra-fine	<b>217</b>	<b>109</b>	109	2,578,177
	1.50 x f	1.49 x f	1.49 x f	3.34 x m

Layout of Grid Blocks that Abut the Surface

wing TE cap



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